	tering to the productive growth and the second to the seco
The state of the s	t new professor to the control of th
	!
N/ Time	•.
N67-28796	
(ACCESSION NUMBER)	(THRU)
<u> </u>	
(PAGES)	(CODE)
<u> </u>	(CATEGORY)
(NASA CR OR TMX OR AD NUMBER)	(CATEGORY)

THE DEVELOPMENT OF DISPERSION-STRENGTHENED NICKEL-BASE CORROSION-RESISTANT ALLOYS

by

STAFF MEMBERS
MELPAR MATERIALS LABORATORY

prepared for

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

	The second secon	
GPO PRICE	\$	CONTRACT NAS 3-7271
CFSTI PRICE(S	3) \$	
Har ∡opy (I	HC)	
Microfiche (M	MF)	

ff 653 July 65

MELPAR, INC.

NOTICE

This report was prepared as an account of Government sponsored work. Neither the United States, nor the National Aeronautics and Space Administration (NASA), nor any person acting on behalf of NASA:

- A.) Makes any warranty or representation, expressed or implied, with respect to the accuracy, completeness, or usefulness of the information contained in this report, or that the use of any information, apparatus, method, or process disclosed in this report may not infringe privately owned rights; or
- B.) Assumes any liabilities with respect to the use of, or for damages resulting from the use of any information, apparatus, method or process disclosed in this report.

As used above, "person acting on behalf of NASA" includes any employee or contractor of NASA, or employee of such contractor, to the extent that such employee or contractor of NASA, or employee of such contractor prepares, disseminates, or provides access to, any information pursuant to his employment or contract with NASA, or his employment with such contractor.

Where trade names or proprietary identifications have been used, they are to be construed as neither an evaluation or an endorsement by NASA. These names have been used solely for describing experimental procedures and/or for indicating equipment or conditions necessary for reproducing the experimental results described in this report.

Requests for copies of this report should be referred to

National Aeronautics and Space Administration Office of Scientific and Technical Information Attention: AFSS-A Washington, D.C. 20546

9<u>&</u>//

SUMMARY REPORT

THE DEVELOPMENT OF DISPERSION-STRENGTHENED NICKEL-BASE CORROSION-RESISTANT ALLOYS

by

Staff Members, Melpar Materials Laboratory

prepared for

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

May 1967

CONTRACT NAS 3-7271

Technical Management
NASA Lewis Research Center
Cleveland, Ohio
Fredric H. Harf, Project Manager
Walter S. Cremens, Research Advisor

MELPAR, INC. 7700 Arlington Boulevard Falls Church, Va.

THE DEVELOPMENT OF DISPERSION-STRENGTHENED NICKEL-BASE CORROSION-RESISTANT ALLOYS

by

Staff Members, Melpar Materials Laboratory

ABSTRACT

The results of a program to produce a dispersion-strengthened 70 nickel-20 chromium-10 molybdenum alloy by vapor plating in a fluidized bed of oxide particles are described. The nickel vapor was obtained by thermally decomposing nickel carbonyl, Ni(CO)₄. The chromium vapor was obtained by decomposing dicumene chromium, $Cr(\vec{C_9}H_{12})_2$, and the molybdenum vapor was obtained by reducing molybdenum pentachloride, MoCl₅. Single and multi-element alloys were made. Metallographic examination of single-element deposit agglomerates of chromium or nickel showed a uniform particle distribution. Production of the triple-element dispersion-strengthened alloys by vapor deposition in a fluidized bed proved unsuccessful because of the low plating rates and difficulties in controlling the chemistry of the product. The triple alloys were characterized by unusually high carbon contents (5 wt percent), owing chiefly to the apparent decomposition of the carbonyl radical and carbon monoxide carrier gas used in vapor plating the nickel. The triple-element matrix alloys were also characterized by the presence of agglomerated oxide particles. These studies indicate that it is possible to vapor plate submicron oxide particles in a fluidized bed with either nickel or chromium and to obtain uniform fine oxide dispersions. The results of adding the nickel to the alloy by an alternative method are also discussed.

THE DEVELOPMENT OF DISPERSION-STRENGTHENED

NICKEL-BASE CORROSION-RESISTANT ALLOYS

by Staff Members Melpar Materials Laboratory

SUMMARY

Attempts were made to produce a 70 nickel—20 chromium—10 molybdenum dispersion-strengthened alloy by plating 0.1-micron-diameter zirconia particles successively with molybdenum, chromium and nickel. This plating sequence was in the order of decreasing decomposition temperature of the precursor compounds. The molybdenum was deposited by decomposing MoCl₅ in the vapor state at 575°C. During this process, the zirconia was observed to agglomerate and sinter into approximately 25-micron-diameter particles. Another major problem in building up the three-element alloy was in the deposition of the nickel. The nickel was deposited by thermally decomposing nickel-tetracarbonyl, Ni(CO)₄. When nickel was deposited on the chromium layer for making the three-element alloy, the following reaction took place:

$$2 CO \longrightarrow CO_2 + C$$

The excess carbon monoxide, CO, came from both the carrier gas and the decomposing nickel carbonyl. Alloys produced in this manner contained more than 5 wt percent carbon.

Attempts were made to add submicron nickel powder to the fluidized bed after plating the molybdenum and the chromium by the vapor deposition process. These procedures did not produce the desired degree of dispersoid distribution and elemental homogeneity, owing partly to the agglomeration of the fine-particle oxide particle during molybdenum deposition.

Vapor plating in a fluidized bed has produced uniform fine oxide dispersions with either nickel or chromium. The fluidized-bed process is capable of producing dispersion-strengthened alloys having good dispersoid distribution, provided the precursor compounds used for plating the elements decompose at low temperatures and that the oxide dispersoids do not agglomerate.

INTRODUCTION

The development of airbreathing engines of higher thrust is limited chiefly by the properties of the materials used for the turbine stator vanes and the rotating turbine buckets. One of the most promising ways to develop alloys with higher use temperatures and stresses is through dispersion strengthening. At present, the most promising dispersion-strengthened materials appear to be such dispersion-strengthened nickel base alloys as have been produced by DuPont and Sheritt-Gordon.

It was a goal of this program to produce a material with a 3,000 hour rupture strength of 15,000 psi at 2000° F. This was to be accomplished with an alloy that combines both dispersion and solid-solution strengthening. An alloy with 2 to 8 vol percent dispersoid was to be produced with a matrix composition of 70 Ni -20 Cr -10 Mo.

The addition of chromium serves two purposes: it greatly increases the oxidation resistance of nickel and, it acts as a solid-solution strengthener of nickel. Molybdenum is one of the most effective solid-solution strengtheners of nickel and was added for this reason. Thus, a dispersion-strengthened alloy with a matrix of nickel containing chromium and molybdenum might meet the stringent mechanical strength and oxidation resistance specifications.

Effective dispersion strengthening requires that the second-phase dispersoid particles be uniformly distributed. The strength of dispersion-strengthened alloys is proportional to $d^{-1/2}$ for coarse particles and d^{-1} for fine particles, (1,2) where d is the mean free path between oxide particles. To optimize strength and ductility, it is necessary to use the smallest particles possible consistent with adequate dispersoid-matrix stability. This permits the smallest mean free path, d, with a low volume fraction of dispersed phase.

To accomplish these goals it was postulated that if fine oxide particles (<0.1 micron diameter) could be uniformly plated with matrix elements, an ideal dispersion could be produced. The mean free path, d, would be minimized and the particle spacing would be uniform. To effect the uniform plating of the particles, a bed of fine oxide particles was fluidized with a gas stream. By decomposing the various volatile metallic precursor compounds in the region of the fluidized particles, the metal(s) could then be plated out on the oxide particles. This vapor plating method had previously been proven feasible for producing dispersion-hardened gold and platinum alloys. (3)

Previous investigators of chemical vapor deposition⁽⁴⁾ have relied almost exclusively upon nickel tetracarbonyl for producing nickel coatings. They showed that molybdenum coatings could be prepared by chemical vapor deposited from both MoCl₅ and the molybdenum carbonyl. The molybdenum carbonyl may be decomposed at lower temperatures (400° vs 650°C) than the molybdenum pentachloride but carbon contamination has been a problem in coatings produced from the carbonyl. For this reason,

molybdenum pentachloride was selected as the precursor compound to produce molybdenum coatings. Chromium coatings have been produced by decomposing chromium iodide, chromium carbonyl, and dicumene chromium⁽⁴⁾. Dicumene chromium decomposes at the lowest temperature of these three materials and was selected for this reason.

It was the purpose of Task I of this program to produce dispersion-strengthened billets of a corrosion-resistant nickel-base matrix containing three levels of each of two oxide dispersoids. The target alloys contained 70 Ni -20 Cr -10 Mo with 2, 5 or 7 volume percent ThO $_2$ or ZrO $_2$. The average oxide dispersoid was to be ≤ 0.02 to 0.10 microns with an interparticle spacing in the matrix of ≤ 0.3 to 1.0 microns. Vapor plating of the oxide dispersoids was to be conducted in a fluidized bed of the dispersoid particles to produce a multilayer coating of Ni, Cr, and Mo. Coated particle agglomeration was to be accomplished by collapsing the fluidized bed and raising the temperature while maintaining a reducing atmosphere. The billets were to be made by isostatically pressing the agglomerated vapor coated dispersoid particles.

Evaluation of the material so produced was to include metallography by both optical and electron microscopy, thermal stability tests and chemical analysis. Linear analysis of the electron photomicrographs was to be used to determine the microstructural parameters. Thermal stability was to be determined by a 100-hour heat treatment in an inert atmosphere at a temperature equivalent to 0.85 of the melting point. Linear analysis was to be carried out after heat treatment to determine any microstructural changes.

It was the purpose of task II of this program to hot extrude the billets produced in task I, and to obtain mechanical property data for the material.

EXPERIMENTAL PROCEDURE

Design of the Fluidized-Bed System

The first objective of this program was to design a system in which the fine particle dispersoids could be fluidized and subsequently vapor plated. The design for the fluidized bed system as shown in figures 1 and 2 evolved from a series of experiments to obtain optimum deposition with minimum blowover. A controlling part of the design is the bottom of the fluidized bed.

The first design shown in figure 3-a proved impractical because the glass frit became plugged by the decomposing compounds. For this reason, a tapered column was used. The taper designs shown in figures 3-b and 3-c permitted the oxide powder to fall into the "T" and cause plugging. By extending the taper section and placing an

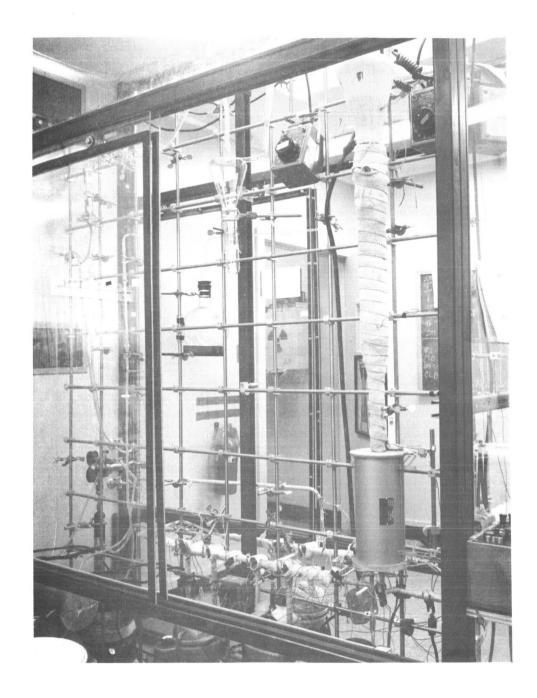


Figure 1. Fluidized-Bed Vapor Plating System for Producing Dispersion-Hardened Alloys

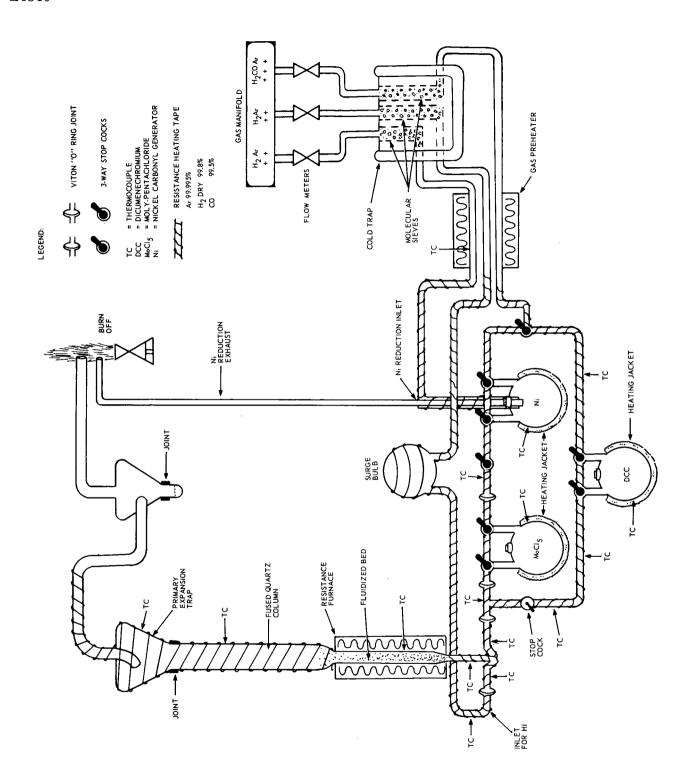
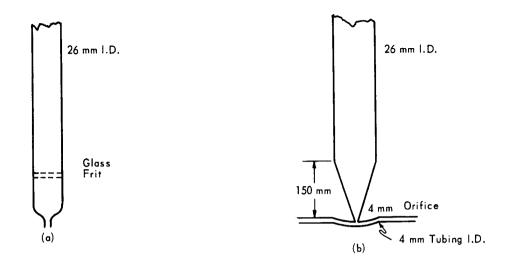


Figure 2. Fluidized-Bed Reactor for Producing Dispersion-Hardened Alloys



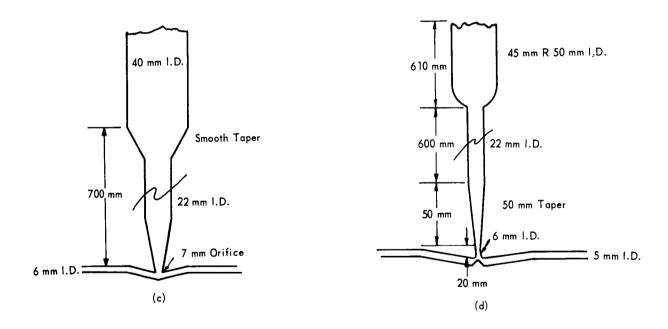


Figure 3. Designs for Bottom Part of Column of Fluidized Bed

undulation at the base of the column, plugging was minimized. This design is shown in figure 3-d. The sizes of the column shown in figure 3-d are optimized and identical with those used in most of the experiments. Orifices larger than 6-mm ID permitted oxide from the bed to run into the "T" and eventually cause plugging.

To minimize blowover, the column was expanded from 45-mm I.D. to 70-mm I.D. above the heated zone. Above the expanded section an inverted Erlenmeyer flask was attached to the column, as shown in figure 1.

It was necessary to heat all of the gas lines with heating tape as shown in figure 1 to prevent condensation of the volatile components.

From the observation of the operation of tapered fluidized beds, the following generalizations were made:

- a. To obtain turbulent fluidization, a critical amount of dispersoid power (for example, 7 g of MgO for a 2-in.-dia. column)⁽⁵⁾ must be used together with a critical flow rate.
- b. A vibrator was necessary to prevent sticking of the powders to the heated walls.
- c. Too large an orifice at the bottom permitted powder to enter the "T", which resulted in plugging.
- d. Too narrow an orifice at the bottom caused the gas flow to channel through the bed instead of fluidizing the bed. The particle bed would ultimately collapse in this condition.

Fluidized Bed Reactor Operation

In actual practice, the operation of the fluidized-bed system shown in figures 1 and 2 was as follows. Gas from cylinders passed through a cold trap and then through a preheat furnace. One line of the gas was fed into one side of the "T" leading to the fluidized bed and served to support the bed of fine particle dispersoids. The other line of gas passed through the precursor compound. After the vapor plating operation was completed, it was routine to raise the temperature of the fluidized bed 50° to 200° C over the decomposition temperature of the last element deposited for 1 to 2 hours. This caused the individual coated dispersoid particles to agglomerate into 25- to 50-micron-diameter composite particles. These large particles were not pyrophoric and could be easily handled in air.

Materials Used for Vapor Plating

<u>Dispersoid Material.</u>— Two dispersoid materials, thoria and zirconia, were used on this program. Both were supplied by Vitro Laboratories. Examination by electron microscopy showed both materials to be less than 0.1 micron in diameter. The average particle size appeared to be 0.02 to 0.05 microns with less than 5 percent of the particles being 0.3-0.7 microns in diameter.

Nickel Plating Materials.— The nickel used for producing the nickel carbonyl was Fisher low cobalt reagent grade -200 mesh. Chemically pure grade carbon monoxide was used for preparing the nickel tetracarbonyl. High-purity dry hydrogen and 99,995 percent pure argon were used in all of the plating runs.

The nickel added directly to molybdenum and chromium plated zirconia was Sheritt-Gordon NF-IM (0.6 to 1.4 micron) powder.

<u>Chromium Plating Materials.</u>— The dicumene chromium was obtained from Union Carbide Chemical Company (90 percent pure).

Molybdenum Plating Materials. — The molybdenum pentachloride obtained from Climax Molybdenum Co. had a batch analysis as follows: Mo 35.1 percent, Cl 63.9 percent, Fe 0.042 percent, Si 0.006 percent, Cu 0.0007 percent, Ni 0.012 percent.

Nickel Deposition

Nickel coatings were produced by the decomposition of nickel tetracarbonyl, Ni(CO)₄. Nickel tetracarbonyl was prepared continuously in the system. Hydrogen gas was passed through high-purity nickel powder, which was heated to 400°C, to reduce any surface oxide and to activate the surface. After reducing with hydrogen for 4 hours, carbon monoxide was passed into the nickel at 125°C and a flow rate of one liter/min. The carbon monoxide reacts readily at this temperature to form nickel carbonyl, Ni(CO)₄.

The nickel carbonyl then passed through feeder lines heated to 100°C and into the fluidized bed column heated to a temperature between 160°C and 210°C. The nickel tetracarbonyl decomposes according to the reaction

$$Ni(CO)_4$$
 Ni + 4 CO

The nickel plates out on the fine particles and the CO passes out of the system to a burn-off. Under these conditions, the plating rate was 0.50 g/hr.

Chromium Deposition

The chromium deposition techniques were developed as part of another investigation, (5)

The chromium was deposited by decomposing dicumene chromium, $Cr(C_9H_{12})_2$. Hydrogen and argon were mixed in 1:1 ratio and passed into a container of dicumene chromium heated to 190° C. The dicumene chromium vapor pressure at this temperatre is 10 torrs. The vapor and carrier gases then passed through feeder lines at 200° C and the vapor was decomposed in the fluidized bed at 325° C. Temperatures higher than this caused excessive breakdown of the dicumene molecule and subsequent C deposition. Using a flow rate of one liter/min through the dicumene chromium, plating rates of 0.5 g/hr were realized.

Lower decomposition temperatures could be used if hydrogen iodide gas was added to the fluidized bed. This was done by bleeding HI at the rate of 0.05 liter/min into the gas stream as shown in figure 2. The HI then mixed with the dicumene chromium vapor in the fluidized bed and catalyzed the decomposition reaction. Temperatures as low as 285°C could be used for chromium deposition when using HI. The resulting coatings, however, contained objectionably high iodine contents. The iodine was present as chromium iodide, CrI_3 . The iodine content varied between 5 and 15 wt percent depending on slight changes in the HI flow rate.

Typical chromium coatings produced without using the HI catalyst contained approximately 3 wt percent C. Using both hydrogen gas and HI catalyst with a HI-to-dicumene chromium ratio of 1:4, the carbon content was reduced to 0.3 wt percent (5).

Electron micrographs of agglomerated chromium coated MgO particles produced by this process without HI are shown in figures 4 and 5. The distribution of the dispersoid particles is good. The chromium platings consisted almost entirely of amorphous material as determined by x-ray examination. When the coated particles were heated to 800° C, the amorphous deposit crystallized and was analyzed to be Cr_{23}C_6 by x-ray diffraction. No metallic chromium lines were observed.

Molybdenum Deposition

The molybdenum was deposited by passing hydrogen and argon in a 1:1 ratio through a pot of molten molybdenum pentachloride, MoCl₅, heated to 170°C. The vapor and carrier gases then passed through feeder lines at 200°C and were decomposed in the fluidized bed at 575°C. By using a flow rate of 0.8 liter/hr through the MoCl₅, plating rates of 0.2 g/hr were realized.

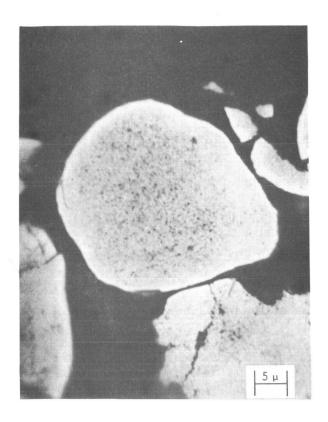


Figure 4. Electron Micrograph of Cr-32% MgO Composite Particle

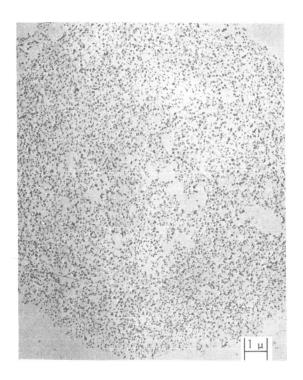


Figure 5. Electron Micrograph of Cr-32% MgO Particle

Multiple-Element Deposition

The individual elements had to be vapor deposited in the fluidized bed in the order of decreasing precursor compound decomposition temperatures (Mo, Cr, Ni). A temperature 50°C above the deposition temperature of Cr (325°C) and Ni (190°C) caused agglomeration of the individual coated particles precluding uniform coating and dispersoid distribution.

RESULTS AND DISCUSSION OF RESULTS

Nickel Deposition

Three runs were made to determine the parameters for nickel deposition. Parameters and results are shown in table 1. The nickel carbonyl generator was maintained at 125° to 165°C. For optimum formation of nickel carbonyl the generator was operated at 125°C. A fluidized bed temperature between 220° and 185°C was used. The use of decomposition temperatures above 210°C caused excessive carbon deposition owing to the decomposition of the carbon monoxide

$$2CO \longrightarrow C + CO_2$$

Decomposition temperatures below 190°C resulted in very low nickel deposition rates owing to apparent incomplete nickel carbonyl decomposition. Under the best conditions (generator 125°C; flow rate 1 liter/min each of carbonyl plus CO, and of argon; bed at 195° - 200°C), plating rates of 0.5 g/hr were realized.

Figure 6 is an electron micrograph of the material produced during run 3. The distribution of the oxide dispersoid particles in the nickel matrix appears to be good.

Molybdenum Deposition

Eight experimental runs were conducted in which molybdenum was plated on zirconia particles. Two runs were stopped after the molybdenum deposition. The other six runs were plated also with chromium or/and nickel. Agglomeration of the fine $\rm ZrO_2$ particles was observed to occur at the $\rm MoCl_5$ decomposition temperature of 575°C. Figure 7 is a photomicrograph of the material produced by plating molybdenum on $\rm ZrO_2$ for 28 hrs. The powder obtained from the collapsed fluidized bed was pressed in a 3/8-inch-diameter die at 100,000 psi and vacuum sintered at 1500°C for 1 hour prior to metallographic preparation. Chemical analysis showed the sample contained 8.9 percent Mo. The molybdenum and the zirconia are segregated. The metallic particles in this structure contained no zirconia particles

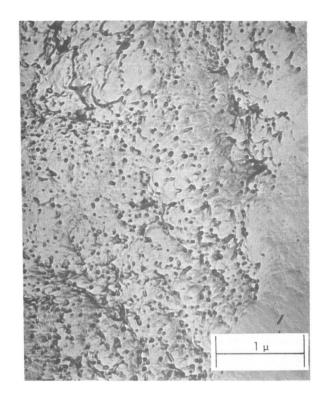


Figure 6. Electron Micrograph of Sintered Ni – ${\rm ZrO}_2$ Particles Produced by Vapor Plating in Run No. 3, Table 1

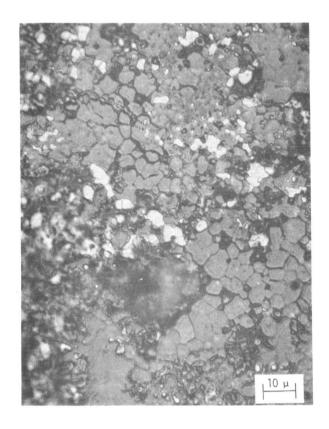


Figure 7. 8.9 Percent Molybdenum- ${\rm ZrO_2}$ After Pressing and Sintering at 1500° C for 1 Hour

TABLE I

RESULTS OF VAPOR PLATING NICKEL ON SUBMICRON

OXIDE PARTICLES

Run Number	1	2	3
Dispersoid	$^{\mathrm{Al}}{_2}^{\mathrm{O}}{_3}$	$^{\mathrm{Al}_2\mathrm{O}}_3$	${f ZrO}_2$
Particle size (4)	0.05	0.05	0.10
Wt. dispersoid (g)	7.76	7.65	5.60
Ni(CO) ₄ generator temperature (°C)	55	60	125
Fluidized bed temperature (°C)	220	200	200
Plating Time (hr.)	5.5	5.0	18
Recovered Wt. (g)	9.01	10.12	9.00
Chemical Analysis			
% Ni	20.9		30.1
% c			0.34
% S	0.010		0.026

when examined by electron microscopy. The zirconia particles apparently agglomerated in the fluidized bed before plating took place. Molybdenum deposition then occured in the presence of the larger agglomerated particles. During the vacuum sintering at 1500°C, the molybdenum subsequently agglomerated into regions as shown in figure 7.

Multiple-Element Depositions

Five runs were made in which first molybdenum and then one other element (Cror Ni) were deposited. These dual-element runs were all terminated before completion because of plugging of the reactor or excessive blowover. Plugging was caused by agglomeration of the particles and resultant fluidized-bed collapse and/or condensation of the precursor compound in the lines near the base of the fluidized bed or in the "T" section. The latter was usually caused by a lowering of the temperature in these parts. Temperature control of the system was difficult owing to the inherent instability of the process and the slow response of the system heaters.

One successful attempt was made to produce a triple-element alloy entirely by vapor plating. The elements were deposited in the order of molybdenum, chromium, and nickel. Using the optimum plating conditions previously described, plating times of 6, 16, and 16 hours were used respectively for Mo, Cr, and Ni. The resultant alloy contained 11.0 percent Mo, 35.9 percent Cr, 12.0 percent Ni, 5.2 percent C, and 35.9 percent ZrO₂ by difference. While it was possible to deposit nickel on zirconia as described in section 4.1 with only a 0.34 percent carbon content, this run produced a higher carbon content. This high carbon content made metallographic examination of the product difficult. The oxide particles clumped as the one shown in figure 8 in material taken from a later run.

Because the high carbon content was apparently related to the depositing of nickel, an attempt was made to add the nickel directly in the form of fine powder. Sheritt Gordon NF-1M, 0.6- to 1.4-micron-diameter nickel powder was added in one run after the molybdenum and chromium had been vapor plated onto the zirconia particles. The nickel was added slowly to the top of the fluidized bed. Mixing was accomplished by the motion of the bed and was continued for 2 hours. The fluidized bed temperature was then raised to 750°C for 2 hours to permit sintering. Figure 8 shows a light micrograph of the product. The oxide phase had segregated into large particles 10 to 20 microns in diameter. This run contained 85.0 percent Ni, 3.4 percent Cr, 4.6 percent Mo, 0.2 percent C, 0.003 percent S and 6.7 percent ZrO₂ by difference.

In another run, the nickel powder was added after plating the molybdenum; the chromium was then plated on the mixture of molybdenum-coated $\rm ZrO_2$ and nickel powder. Metallographic examination showed that the oxide particles were segregated as in the previous run. This run contained 85.0 percent Ni, 3.4 percent Cr, 4.6 percent Mo, 0.21 percent C, and 6.8 percent $\rm ZrO_2$ by difference.



Figure 8. Large Clump of Zirconia in Ni-Cr-Mo Alloy

CONCLUSIONS AND RECOMMENDATIONS

Vapor plating fine oxide particles in a fluidized bed has produced nickel or chromium with apparently good oxide particle distribution. Samples so produced contain a high volume fraction of dispersoid (> 40 percent), but are characterized by high carbon content.

Fine zirconia particles (0.1-micron diameter) at the temperature necessary for molybdenum pentachloride reduction, 575°C, agglomerated and sintered into 25-to 50-micron-diameter particles. Unless molybdenum can be deposited at a lower temperature, it seems impossible to produce dispersion-strengthened molybdenum or any alloy containing molybdenum with a uniform dispersoid distribution by the fluidized-bed technique. While these experiments were conducted on zirconia particles, it is reasonable to suspect that other oxides will similarly agglomerate during molybdenum vapor deposition.

A possible solution to the molybdenum deposition problem would be the use of molybdenum carbonyl, Mo(CO)₆ which could be decomposed as low as 400°C. But this process has an inherently high resultant carbon content.

The fluidized-bed process is capable of producing alloys of good dispersoid particle size and distribution provided that the precursor compounds can be applied at relatively low temperatures. Other methods of plating the dispersoid with metals, such as evaporation, sublimation and exploding wires of the elements, might be used inside the fluidized bed. These methods require no external heating of the fluidized-bed particles, thus helping to prevent agglomeration and sintering. Another advantage of using elemental sources would be the elimination of impurities that are picked up from the decomposing precursor compounds.

REFERENCES

- 1. G. S. Ansell and F. V. Lenel, Criteria for Yielding of Dispersion-Strengthened Alloys, Acta Met. 8,612 (Sept. 1960).
- 2. G. S. Ansell, <u>Mechanical Properties of Two-Phase Crystalline Materials</u>, International Symposium on High Temperature Technology, Asilomar, Calif., 1963.
- 3. M. L. Gimpl and N. Fuschillo, <u>Research on Electrical Conductors for High Temperature Applications</u>, AFML-TR-66-171 (1966).

- 4. "Vapor Deposition" ed. Carroll F. Powell, Joseph H. Oxley and John M. Blocher, John Wiley (1966).
- 5. M. L. Gimpl and N. Fuschillo, <u>Development of Dispersion Hardened</u> Chromium, Final Report, Contract No. NAS3-7608, NASA CR 54612 (1966).

DISTRIBUTION LIST

(The number in parentheses is the number of copies sent to each addressee)

NASA Headquarters (1) 600 Independence Avenue Washington, D. C. 20546 Attn: N. F. Rekos (RAP)

NASA Headquarters (1) 600 Independence Avenue Washington, D. C. 20546 Attn: G. C. Deutch (RRM)

NASA Headquarters (1) 600 Independence Avenue Washington, D. C. 20546 Attn: R. H. Raring (RRM)

NASA-Lewis Research Center (1) 21000 Brookpark Road Cleveland, Ohio 44315 Attn: G. M. Ault, M.S. 105-1

NASA-Lewis Research Center (1) 21000 Brookpark Road Cleveland, Ohio 44135 Attn: Technology Utilization Office, M.S. 3-19

NASA-Lewis Research Center (1) 21000 Brookpark Road Cleveland, Ohio 44135 Attn: Patent Counsel, M.S. 501-3

NASA-Lewis Research Center (1) 21000 Brookpark Road Cleveland, Ohio 44135 Attn: I. I. Pinkel, M.S. 5-3

NASA-Lewis Research Center (2) 21000 Brookpark Road Cleveland, Ohio 44135 Attn: Library, M.S. 60-3

NASA-Lewis Research Center (1) 21000 Brookpark Road Cleveland, Ohio 44135 Attn: Report Control Office M.S. 5-5 NASA-Lewis Research Center (1) 21000 Brookpark Road Cleveland, Ohio 44135 Attn: Paul Hacker, M.S. 5-3

NASA Lewis Research Center (1) 21000 Brookpark Road Cleveland, Ohio 44135 Attn: J. Howard Childs, M.S. 60-4

NASA-Lewis Research Center (5) 21000 Brookpark Road Cleveland, Ohio 44135 Attn: F. H. Harf, M.S. 60-6

NASA-Lewis Research Center (1) 21000 Brookpark Road Cleveland, Ohio 44135 Attn: A. E. Anglin, M.S. 60-6

NASA-Lewis Research Center (1) 21000 Brookpark Road Cleveland, Ohio 44135 Attn: John H. DeFord, M.S. 60-5

NASA-Lewis Research Center (1) 21000 Brookpark Road Cleveland, Ohio 44135 Attn: J. W. Weeton, M.S. 49-1

NASA-Lewis Research Center (1) 21000 Brookpark Road Cleveland, Ohio 44135 Attn: Dr. W. S. Cremens, M.S. 49-1

FAA Headquarters (1) 800 Independence Avenue, S. W. Washington, D. C. 20553 Attn: Brig. Gen. J. C. Maxwell

FAA Headquarters (1) 800 Independence Avenue, S. W. Washington, D. C. 20553 Attn: F. B. Howard SS/210 NASA Scientific and Technical (6)
Information Facility
P. 0. Box 5700
Bethesda, Maryland 20014
Attn: NASA Representative
RQT-2448

U. S. Atomic Energy Commission (1) Washington, D. C. 20545 Attn: Technical Reports Library

U. S. Atomic Energy Commission (1) Washington, D. C. 20545 Attn: Jules Simmons

Oak Ridge National Laboratory (1) Oak Ridge, Tennessee 37830 Attn: M. L. Picklesimer

Air Force Office of Scientific (1)
Research
Propulsion Research Divison
USAF Washington, D. C. 20525
Attn: Dr. M. Slawsky

Defense Documentation Center (DDC) Washington, D. C. 20525 Cameron Station (1) Attn: T. F. Kearns 5010 Duke Street Alexandria, Virginia 22314 NASA-Langley Research C

Headquarters (1)
Wright Patterson AFB, Ohio 45433
Attn: MAMP: C. Lombard

Headquarters (1)
Wright Patterson AFB, Ohio 45433
Attn: MATB: G. A. Gegel

Headquarters (1) Wright Patterson AFB, Ohio 45433 Attn: MATB: Lt. A. Lopez

Headquarters (1)
Wright Patterson AFB, Ohio 45433
Attn: MAAM: Technical Library

Headquarters (1)
Wright Patterson AFB, Ohio 45433
Attn: AFSC-FTDS: SM Sgt.
J. C. Ingram, Jr.

Headquarters (1)
Wright Patterson AFB, Ohio 45433
Attn: AFML: Dr. A. M. Lovelace

Headquarters (1)
Wright Patterson AFB, Ohio 45433
Attn: Directorate of Materials

Headquarters (1)
Wright Patterson AFB, Ohio 45433
Attn: SESOS: J. L. Wilkins

Headquarters (1) Wright Patterson AFB, Ohio 45433 Attn: MAMP: I. Perlmutter

Department of the Navy (1) ONR Code 429 Washington, D. C. 20525 Attn: Dr. R. Roberts

U. S. Army Aviation Materials (1)
Laboratory
Fort Eustis, Virginia 23604
Attn: John White, Chief, SMOFE-APG

Chief, Bureau of Naval Weapons (1)
Dept. of the Navy
Washington, D. C. 20525
Attn: T. F. Kearns

NASA-Langley Research Center (1) Langley Field, Virginia 23365 Attn: Library

NASA-Langley Research Center (1) Langley Field, Virginia 23365 Attn: Richard Pride

NASA-Marshall Space Flight Center (1) Huntsville, Alabama 35812 Attn: Library

Jet Propulsion Laboratory (1) 4800 Oak Grove Drive Pasadena, California 91102 Attn: Library

Army Materials Research Agency (1) Watertown Arsenal Watertown, Massachusetts O2172 Attn: S. V. Arnold, Director

NASA-Ames Research Center (1) Moffett Field, California 94035 Attn: Library NASA-Goddard Space Flight Center Greenbelt, Maryland 20771 (1) Attn: Library

NASA-Goddard Space Flight Center Greenbelt, Maryland 20771 (1) Attn: D. F. Hasson Code 714

NASA-Goddard Space Flight Center Greenbelt, Maryland 20771 (1) Attn: C. E. Vest Code 249.1

NASA-Manned Space Flight Center Houston, Texas 77058 (1) Attn: Library

NASA-Manned Space Flight Center Houston, Texas 77058 (1) Attn: N. Chaffee E. B. 4

NASA-Flight Research Center (1) P. O. Box 273 Edwards, California 93523 Attn: Library

Aerospace Corp. (1) Technical Documents Group P. O. Box 95085 Los Angeles, California 90045

Aerojet-General (1) Nucleonics Division San Ramon, California 95483 Attn: Mr. J. S. Luce

American Society for Metals (1) Metala Park Novelty, Ohio 44073 Attn: Dr. Taylor Lyman

Atlantic Research Corp. (1)
Shirley Memorial Hwy.-Edsel Road
Alexandria, Virginia
Attn: J. F. Hoeble

AVCO Corp. (1)
201 Lowell Street
Wilmington, Massachusetts 01887
Attn: Dr. Neil Rogen, Chief,
Metals and Ceramics

Defense Metals Information Center (DMIC) (1)
Battelle Memorial Institute
505 King Avenue
Columbus, Ohio 43201

Battelle Memorial Institute (1) 505 King Avenue Columbus, Ohio 43201 Attn: Dr. R. I. Jaffee

Battelle Memorial Institute (1) 505 King Avenue Columbus, Ohio 43201 Attn: E. S. Hodge

Battelle Memorial Institute (1) 505 King Avenue Columbus, Ohio 43201 Attn: Dr. B. Wilcox

Cobalt Information Center (1)
Battelle Memorial Institute
505 King Avenue
Columbus, Ohio 43201

The Bendix Corp. (1)
Research Laboratories Div.
Southfield, Michigan 48075
Attn: C. B. Sung

Brush Beryllium Corp. (1) 17876 St. Clair Avenue Cleveland, Ohio 44112 Attn: Mr. W. W. Beaver

Chromalloy Corp. (1)
Sintercast Div.
169 Western Highway
West Nyack, New York 10994
Attn: J. L. Ellis

Case Institute of Technology (1) University Circle Cleveland, Ohio 44106 Attn: Prof. L. Leonard

Curtiss-Wright Corp. (1)
Metals Processing Div.
760 Northland Ave.
Buffalo, New York 14215
Attn: B. Triffleman

Denver Research Institute (1) University Park Denver, Colorado 80210 Attn: Library

Boeing Co. (1) P. O. Box 733 Renton, Washington 98055 Attn: W. E. Binz, SST Unit Chief Dow Metal Products Div. (1) Dow Chemical Co. Hopkins Bldg. Midland, Michigan 48640 Attn: H. Buckelew

Douglas Aircraft Co-MSFD (1) 3000 Ocean Park Blvd. Santa Monica, California 90406 Attn: Dr. D. H. Killpatrick A-260

Fansteel Metallurgical Corp. (1) Number One Tantalum Place North Chicago, Illinois 60064 Attn: P. C. Rossin

Firth Sterling Inc. (1)
Powder Metals Research
P. O. Box 71
Pittsburgh, Pennsylvania 15230

Falconbridge Nickel Mines, Ltd. 7 King Street, East (1)
Toronto, Ontario, Canada
Attn: L. G. Bonar

General Electric Co. (1)
Materials and Processes Lab.
Schenectady, New York 12305
Attn: C. T. Sims, Bldg. 55

General Electric Co. (1)
Advanced Technology Lab.
Schenectady, New York 12305
Attn: Library

General Electric Co. (1)
Materials Devel. Lab. Oper.
Advance Engine and Tech. Dept.
Cincinnati, Ohio 45215
Attn: L. P. Jahnke

General Motors Corp. (1)
Allison Div.
Indianapolis, Indiana 46206
Attn: D. K. Hanink, Materials
Lab.

General Technologies Corp. (1) 708 North West St. Alexandria, Virginia 22314 Attn: James C. Withers E. I. DuPont de Nemours and Co., (1)
Inc.
Pigments Dept Metal Products
Wilmington, Delaware 19898
Attn: Dr. Warren I. Pollock

Ilikon Corp. (1)
Natick Industrial Center
Natick, Massachusetts
Attn: Dr. L. J. Bonis

International Nickel Co. (1) 67 Wall St.
New York, New York 10005
Attn: R. R. Dewitt

IIT Research Institute (1) Technology Center Chicago, Illinois 60616 Attn: S. L. Blum

International Nickel Co. (1)
P. D. Merica Research Lab.
Sterling Forest
Suffern, New York 10901
Attn: Dr. F. Decker

Ladish Co. (1)
Government Relations Div.
Cudahy, Wisconsin 53110
Attn: C. Burley, Jr.

Arthur D. Little, Inc. (1) 20 Acorn Park Cambridge, Massachusetts Attn: Dr. B. Bovarnick

P. R. Mallory and Co., Inc. (1) 3029 E. Washington St. Indianapolis, Indiana 46206 Attn: Technical Library

Lockheed-Georgia Company (1) Research Laboratory Marietta, Georgia 30060 Attn: Dr. W. S. Cremens

Lockheed Palo Alto Research (1)
Labs.
Materials and Science Lab. 52-30
3251 Hanover St.
Palo Alto, California 94304
Attn: Technical Infor. Center

Lockheed Palo Alto Research (1)
Labs.

Materials and Science Lap 52-30 3251 Hanover St. Palo Alto, California 94304 Attn: Dr. C. G. Goetzel

Lockheed Palo Alto Research (1)

Materials and Science Lap 52-30 3251 Hanover St.
Palo Alto, California 94304
Attn: E. C. Burke

Massachusetts Institute of (1)
Techn.
Matallana Parantment DW 0.70

Metallurgy Department, RM 8-305 Cambridge, Massachusetts 02139 Attn: Prof. N. J. Grant

Melpar, Inc. (1) 3000 Arlington Blvd. Falls Church, Virginia 22903 Attn: Dr. N. Fushillo

Fordon McKay Lab. (1) 6 Oxford St. Cambridge, Massachusetts O2138 Attn: Dr. M. Ashby

Mitron Corp. (1) 899 Main St. Waltham, Massachusetts 02154

Narmco Research and Development Div. (1) Whittacker Corp. 3540 Aero Court San Diego, California 92123 Attn: Dr. F. J. Riel, Tech. Dir.

N.R.A. Inc. (1) 35-01 Queens Blvd. Long Island City, N.Y. 11101 Attn: Dr. S. Grand

North Star Research and (1)
Development Inst.
3100 Thirty Eight Ave., South
Minneapolis, Minnesota 55406
Attn: Dr. J. W. Clegg

Nuclear Materials Company (1) West Concord, Mass. 01781 Attn: Dr. H. Levingston

Ohio State University (1)
Columbus, Ohio 43210
Attn: Prof. M. G. Fontana,
Chairman, Dept. of
Metallurgical Eng.

Raytheon Co. (1)
Microvane and Power Tube Div.
Waltham, Massachusetts O2154
Attn: Mr. G. Freedman

Rensselaer Poly Tech. (1) Troy, New York 12180 Attn: Prof. Fritz V. Lenel

Rensselaer Poly Tech. (1) Troy, New York 12180 Attn: Prof. G. S. Ansell

Sherritt Gordon Mines, Ltd. (1) Research and Development Div. Fort Saskatchewan, Alberta, Canada Attn: Dr. L. F. Norris

Sherritt Gordon Mines, Ltd. (1) Research and Development Div. Fort Saskatchewan, Alberta, Canada Attn: Dr. V. H. Machiw

Sherritt Gordon Mines, Ltd. (1) Research and Development Div. Fort Saskatchewan, Alberta, Canada Attn: Dr. J. I. Evans

Solar Div. (1)
International Harvester Corp.
San Diego, California 92112
Attn. J. V. Long, Dir. of
Research

Texas Instruments, Inc. (1)
Materials and Controls Div.
13500 North Central Parkway
Dallas, Texas 75222
Attn: Gene Wakefield

Sylvania Electric Products, Inc. (1) Chemical and Metallurgical Div. Towanda, Pennsylvania 18848 Attn: Dr. J. S. Smith Teledyne Materials Research Co. 303 Bear Hill Road (1) Waltham, Mass. 02154 Attn: Dr. R. Widmer

TRW Electromechanical Div. (1)
TRW Inc.
23555 Euclid Avenue
Cleveland, Ohio 44117
Attn: Dr. A. S. Nemy

TRW Electromechanical Div. (1)
TRW Inc.
23555 Euclid Avenue
Cleveland, Ohio 44117
Attn: Dr. R. Quigg

TRW Electromechanical Div. (1) TRW Inc. 23555 Euclid Avenue Cleveland, Ohio 44117 Attn: G. Doble

Stanford University (1)
Palo Alto, California 94305
Attn: Prof. Oleg Sherby, Dept.
of Mat'l Science

United Aircraft Corp. (1) Pratt and Whitney Div. West Palm Beach, Florida 33402 Attn: J. Moore

Universal-Cyclops Steel Corp. (1) Bridgeville, Pennsylvania 15017 Attn: C. P. Mueller

Vitro Laboratories (1) 200 Pleasant Valley Way West Orange, New Jersey 07052 Attn: Dr. H. McCullough Wah Chang Corp. (1)
Albany, Oregon 97321
Attn: S. Worster

Westinghouse Electric Corp. (1) MacArthur Avenue Bloomfield, New Jersey 07003

United Aircraft Corp. (1) 400 Main St. East Hartford, Connecticut 06108 Attn: Research Library

United Aircraft Corp. (1)
400 Main St.
East Hartford, Connecticut 06108
Attn: E. F. Bradley, Chief,
Mat'ls Eng.

Union Carbide Corp. (1) Stellite Div. Technology Dept. Kokomo, Indiana 46901 Attn: J. J. Phillips

Union Carbide Corp. (1) Stellite Div. Technology Dept. Kokomo, Indiana 46901 Attn: Technical Library

University of California (1) Los Angeles, California 90024 Attn: Dr. G. Hoffman